Rapidity Optimization of the Aircraft Carrier Based on the Minimum Resistance

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Abstract—On the basis of the model of the aircraft carrier "Kitty Hawk", the designer will obtain different models through parody transformation to change its aspect ratio(L_{ω}/B_{ω}), block coefficient(C_b) and displacement(Δ) and calculate the resistance of different models. With the principle of the optimal speed, taking an overall consideration of initial stability, scakeeping performance and the restrictions of flight deck and hangar width of the aircraft carrier, we get a new ship form and the corresponding speed. Compared the results with the parent ship, the speed of the optimized ship has increased from 32 kn to 35 kn. The conclusion is that the rapidity of the new ship has been optimized.

Keywords-aircraft carriers; parody transformation; rapidity; initial stability; seakeeping performance

I. INTRODUCTION

As a highly mobile marine airfields and naval bases, aircraft carrier has received the world attention of the global navy since its creation. To seize command of the sea and air supremacy is a very important task for large aircraft carriers. And the rapidity of the aircraft carrier largely determines the status and combat mission of an aircraft carrier in the war^[1].Improving the rapidity of "Kitty Hawk" with the premise of other performance unchanged has a significance on the development of future aircraft carriers.

II. THE INFLUENCE OF RAPIDITY BY THE ASPECT RATIO (L_{ω}/B_{ω}) CHANGING WITH A CONSTANT DISPLACEMENT

A. The Influence of Rapidity by the Aspect Ratio(L_{ω}/B_{ω})

For the medium and high speed ship, with the change of the length and the corresponding speed, it will get the best length L_{opt} of the lowest overall resistance under the condition of the constant displacement^[2].Increasing the length may reduce the overall resistance of the ship within the preferred length range. According to the statistics, the L_{ω}/B_{ω} value is in the range of 7.10-8.10^[3], which can ensure the model displacement unchanged, and parody transforming Fang Xin-yue College of Ship-building Engineering Harbin Engineering University, HEU Harbin, China 83172711@163.com

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the length and width of the ship is showed in the following schemes Table 1.

According to the five model schemes in table 1, the designer calculated the resistance curve under the speed of 32 kn, as shown in Figure 1 which can conclude that the ship resistance getting smaller as L_{ω}/B_{ω} increasing within the range of 7.20-8.10 and the constant displacement.



Figure 1. L_{ω}/B_{ω} resistance curve of designed speed

B. The minimum B_{ω} limitation by flight deck width

As a special ship that can carry a large number of carrierbased aircrafts, the aircraft carrier needs the consideration of the influence on the stability, cabin layout, the aircrafts of hangar placed and the aircrafts of flight deck placed due to the change of L_{ω}/B_{ω} . Therefore, the fight deck arrangement restricts the transformation of main dimension.

The minimum waterline width and the flight deck width is relevant, the ratio of the waterline width of the foreign angled deck and the flight deck width is about 0.525^[4]. The width of flight deck is determined by the landing runway width and the stopped area of the carrier. And the minimum waterline width should be determined by the width of tail landing runway.

Taking into account the tail landing runway arrangement under the navigation, the tail landing runway should be placed at least 4 folded F-18^[5], which is showed in Figure 2.The width of folded F-18 is 8.39 m^[6]. Abiding by the static park space requirements of the carrier-based aircrafts placed on the flight deck, the safe distance of the carrier-based aircrafts parked on the technology position in park area, is not less than 750mm^[7]. As for "Kitty Hawk", the width of the tail landing runway is 37.9m, and the width of the flight deck is 72.3m. So as for the optimized ship, the width of the flight deck is 68.3m through proportionate relationships. The ratio of the waterline width and the flight deck width of "Kitty Hawk" is 0.55. Therefore, the optimized B_m is 37.2m^[8].



Figure 2. The aircraft arrangement on the tail landing runway under the navigation state

C. The minimum B_{ω} restriction by the hangar width

The width of the hangar, relating to the following factors, such as the size of the aircraft carrier and overall layout, is about 72-80% of waterline width. The increase of the width of the hangar will lead to reduce the leaving space between the side wall and the side of the hangar. However, the ratio of the hangar width and B_{ω} of Kitty Hawk is 0.827, which has surpassed the statistical range of the hangar and the waterline width^[9].

Compared with the waterline width of the Nimitz Class aircraft carrier, the width of Kitty Hawk increased by 1.4m, and the hangar width only increased by 0.4m, so the ratio of the hangar width and the waterline width is 0.808. The change of the aircraft positions or placing ways in hangar from Forrestal-class to Kitty Hawk-class made the hangar width of Kitty Hawk set necessarily at 32.6m, which is considered as the minimum waterline width. By the consideration of the rapidity and hanger placement for Kitty Hawk, the selection of the waterline length and the width is the best value. The waterline width is 39.4m.

scheme	transformation rule	$L_{\omega}(\mathbf{m})$	$B_{\omega}(\mathbf{m})$	L_{ω}/B_{ω}	draft (m)	$\Delta(t)$
scheme A1	length minus 10 meters	291.2	40.4	7.20	11.4	80500
scheme A2	length minus 5 meters	296.2	39.7	7.46	11.4	80589.4
scheme A3	parent ship	301.804	39.4	7.66	11.4	80589.4
scheme A4	length plus 5 meters	306.88	38.7	7.93	11.4	80589.4
scheme A5	length plus 10 meters	310.23	38.3	8.10	11.4	80590

TABLE I. SHIP TRANSFORMATION SCHEME WITH CONSTANT DISPLACEMENT

III. THE INFLUENCE ON RESISTANCE BY BLOCK COEFFICIENT CHANGE

The selection of the block coefficient influences so much on the resistance[10]. So the designer needs to consider the influence on overall resistance by the block coefficient variation during the aircraft carrier rapidity optimization. The model schemes are showed in Table 2.

We calculated the resistance and the powers of the five models (in Table 2) under different speed through the parody transformation and obtained the following curves (Figure 3):



It is known from Figure 3 that the best ship resistance contributes to the minimum block coefficient. The ship resistance increased significantly with the increasing of the block coefficient.

IV. THE INFLUENCE ON THE POWER BY THE DISPLACEMENT CHANGE

Without any consideration of the main dimension influence by the placement of the hangar and the flight deck, we obtained the different ship models by the parody transformation to the parent ship Kitty Hawk. L_{ω}/B_{ω} is selected in the range of 7.66-7.82. And the draft is selected within 3.45-3.77 which is the range of the ratio of the waterline width and the mean draft. The schemes are shown in Table.3.

We calculated the resistance and power of the six models in Table 3 and drew power curves under different speeds.



Figure 4. Speed power curve of different tonnage aircraft carrier

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From the Figure 4 can be drawn the conclusion: when the speed is slightly greater than 35 kn, all the ton power curves are converging to one point. And with the increasing

of speed, the power required by the small-ton class vessel is greater than the power of large-ton class ship.

scheme	C_{b}	$L_{\omega}(\mathbf{m})$	$B_{\omega}(\mathbf{m})$	draft(m)	$\Delta(t)$
scheme B1(parent ship)	0.58	301.804	39.4	11.4	80589.4
scheme B2	0.59	301.804	39.4	11.4	81978.87
scheme B3	0.6	301.804	39.4	11.4	83368.35
scheme B4	0.61	301.804	39.4	11.4	84757.82
scheme B5	0.62	301.804	39.4	11.4	86147.29

BLE II. THE SCHEME OF BLOCK COEFFICIENT VARIATION IN SAME	SHIP
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	transformation rule	L_{ω}/B_{ω}	<i>T</i> (m)	$L_{\omega}(\mathbf{m})$	$B_{\omega}(\mathbf{m})$	$\Delta(t)$	B_{ω}/T
scheme 1	50,000 tons	7.66	10.2	268.1	35.0	53982	3.43
scheme 2	60,000 tons	7.66	10.6	276.1	36.0	62000	3.40
scheme 3	70,000 tons	7.75	11.0	290.1	37.4	73506	3.40
scheme 4	80,000 tons	7.76	11.4	301.8	39.4	80589	3.46
scheme 5	90,000 tons	7.77	11.4	317.2	40.8	90734	3.58
scheme 6	100,000 tons	7.70	11.6	331.2	43.0	105118	3.70

TABLE III.	SCHEME OF DIFFERENT TON DISPLACEMENT SH	HP

V. PERFORMANCE ANALYSIS OF OPTIMIZED SHIP

The best resistance performance depends on the maximum L_{ω}/B_{ω} value within the statistical range through the analysis of main dimensions. Therefore the L_{ω}/B_{ω} value of the optimized ship is 8.10, that is, $L_{\omega}=8.10B_{\omega}$ and the minimum waterline width is 39.4m. Increasing the draft can improve the performance of the propeller with the consideration of the rapidity. Taking the waterline width and draft ratio as 3.45, the $B_{\omega}=3.45T$ and the block coefficient C_b is selected as the minimum value 0.58 within the statistical range.

The load displacement calculation formula is displayed below:

$$\Delta = L_{\omega} \times B_{\omega} \times T \times C_b \times 1.025$$

The size of molded depth reflected the freeboard size at a certain draft, so we determined that the optimized molded depth is 29.7m according to the molded depth and draft ratio from statistics.

The length of the ship is 319.2m(1047 feet) in the the optimized scheme. The new ship speed is between 32.4kn

and 35.6kn when the new ship speed ratio is under the range from 1.0 to 1.1. Considering the economic speed is 35kn, which is obtained from the displacement optimization, the maximum speed of the new ship is taken as 35kn.

The comparison of principal dimensions between the optimized ship type and the parent pattern is showed in Table 4, and the comparison of total factors between optimized ship and parent pattern is showed in Table 5.

By checking the new ship's initial stability and seakeeping, the results are shown in Table 6.

From the Table 6: speed is optimized for the new ship, maximum speed increased from 32 to 35 knots, the displacement increased approximately by over 5,000 tons, and five aircrafts in the hangar are added, and the effectiveness of the aircraft carrier is improved. By comparison with Kitty Hawk, there is no change for the initial stability and the seakeeping.

We calculated out the ship resistance of the optimized ship. Figure 5 is the comparison chart of the power resistance curves for Kitty Hawk and the optimized ship.

TABLE IV. THE COMPARISON OF PRINCIPAL DIMENSIONS BETWEEN OPTIMIZED SHIP TYPE AND PARENT PATTERN

(1)

	$L_{\omega}(\mathbf{m})$	$B_{\omega}(\mathbf{m})$	<i>T</i> (m)	<i>D</i> (m)	C_{b}	$\Delta(t)$	speed (kn)
optimized ship	319.2	39.4	11.4	29.7	0.58	85411	35
Kitty Hawk	301.8	39.4	11.4	29.7	0.58	80588	32

TABLE V. THE COMPARISON OF TOTAL FACTORS BETWEEN OPTIMIZED SHIP AND PARENT PATTERN

	hangar length	hangar width	flight deck length	flight deck width
	(m)	(m)	(m)	(m)
optimized ship	238.6	32.6	330.6	72.3
Kitty Hawk	225.6	32.6	312.6	72.3

TABLE VI	THE COMPARISON OF ESTIMATION FORMULA RESULTS
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	Kitty Hawk	optimized ship
Natural rolling period	23.5(s)	23.5(s)
Heave natural period	8.10(s)	8.10(s)
Initial metacentric height	3.09(m)	3.09(m)

From figure 5 we can see, when the speed of the optimized form reached in the maximum of 35 kn, the power is less than the parent form, so the new ship was optimized in the speed.



Figure 5. The comparison chat between Kitty Hawk and Optimization of ship speed power

VI. CONCLUSION

When the displacement was constant, under the design speed, the resistance decreased with the increase of L_{ω}/B_{ω} , the speed can be optimized by increasing L_{ω}/B_{ω} of the Kitty Hawk. But as the most special ship, some factors need to be taken into account, such as how a hangar carrier in the flight deck was put and the returns of flight deck, the limits of the designed line width as well. Through the analysis, it is thought that the minimum designed line width is 39.4 meters. In the premise of what water length and width keeps as the same, the resistance will increase with L_{ω}/B_{ω} increasing. By guaranteeing L_{ω}/B_{ω} in the statistical range, transform displacement of the Kitty Hawk form, when speed is 35 kn, the influence on the change of displacement in power is very small.

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