

# Improved ADRC of Airborne Electro-optical Stabilized Platform

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**Abstract**—This paper proposed a method for parameter setting of traditional nonlinear extended state observer using Back-Propagation neural network. This method simplified the complicated nonlinear extended state observer parameter setting process. This paper simulated the active disturbance-rejection control system combining Back-Propagation neural network. The simulation results show that this method can significantly enhance the disturbance isolation of the airborne optoelectronic stabilized platform, and the control effect is obviously superior to the traditional active disturbance-rejection control method, which is of great significance to improve the axis stability accuracy of the airborne optoelectronic stabilized platform.

**Keywords**—component; optoelectronic stabilized platform; extended state observer; active disturbance-rejection control; back-propagation neural network; disturbance compensation

## I. INTRODUCTION

As the "eye" of an aircraft, the optoelectronic stabilized platform has the functions of capturing, tracking and aiming at the target. Pointing detectors will be interfered by various disturbances during the flight, which will affect the accuracy of line of sight stabilization. To improve the disturbance isolation can improve the accuracy of line of sight(LOS) stabilization, and makes the image of the optoelectronic stabilized platform clearer and the information acquisition more accurate[1,2].

The traditional control strategy achieves the effect of isolating disturbance by increasing the open-loop cutoff frequency of the system. However, because the open-loop cut-off frequency of the system is strictly limited by the mechanical resonance frequency, the gain of the system in the disturbance frequency band can't be further improved[3]. This limits the improvement in stability accuracy.

Active Disturbance-rejection Control (ADRC) uses disturbance compensation control to suppress disturbance effects. Extended state observer(ESO) is the core of ADRC, the "total disturbance" in the system is observed and compensated by ESO in real time, which greatly improves the disturbance isolation and improves the stability accuracy of the optoelectronic stabilized platform [4,5].

The parameter tuning of ESO is very important to the performance of the observer, Dr. Gao Zhiqiang from Cleveland State University proposed a single parameter [6] design method based on bandwidth concept. These methods have better robustness and can be adapted to various models after adjustment.

The intelligent algorithm with adaptive parameters tuning can improve system performance while maintaining good robustness of the system. Some literatures have tried to adjust parameter of ADRC using the method similar to the method of BP neural network tuning PID [7]. However, most of them only focus on two parameters of Non-linear state error feedback (NLSEF) [8, 9]. In fact, the parameter tuning method of the ESO is more complex. At present, the empirical formula is mainly used to adjust it because the accurate model of the system is difficult to obtain. This method is not only time-consuming, but also difficult to determine the optimal parameters. Therefore, the adaptive adjustment of BP neural network is of great significance for improving the performance of ESO.

## II. THE THEORY OF ADRC

ADRC was first proposed in the late 1980s by Han Jingqing, a researcher at the Chinese Academy of Sciences. Its core idea is "active disturbance rejection". All the external disturbances and the "uncertain dynamics" of the system are called "total disturbance" and the "total disturbance" is estimated in real time by the ESO, to realize the direct feed forward compensation control for disturbance.

ADRC includes three parts: Tracking Differentiator (TD), ESO and NLSEF control law.

### A. Tracking Differentiator

The main function of the TD is to arrange the appropriate transition process according to the control target and the object's endurance, resolving the contradiction between rapidity and overshoot of the system. The steepest TD shown in equation (1) is used in this paper.

$$\begin{cases} x_1(k+1) = x_1(k) + hx_2 \\ x_2(k+1) = x_2(k) + hf \\ fh = fhan(x_1(k) - v(k), x_2(k), r, h) \end{cases} \quad (1)$$

in this equation,  $x_1$  is the tracking value of instruction, and  $x_2$  is the tracking value of instruction differential.  $fhan(x_1(k) - v(k), x_2(k), r, h)$  is the system's fastest control synthesis function, and its specific algorithm as shown in the formula (2):



while the ESO parameters are not within this range, so ESO parameters tuning using BP neural network should adjust the output values of the BP neural network, then use the adjusted values as ESO parameters. Because the BP neural network Corrects weight coefficients according to the gradient descent method, the disadvantage of this method is that if the output value and ideal output deviation is too large it is easy to diverge, therefore need to select the suitable coefficient of the output layer, make the neural network better convergence.

### B. A Brief Introduction of BP Neural Network for Tuning ESO Parameters

The output of the neural network input layer as defined above is:

$$O_i^{in} = x(i), i = 1, 2, 3, 4 \quad (5)$$

The inputs and outputs of the hidden layer are:

$$\begin{cases} Net_j^{im}(k) = \sum_{i=1}^4 w_{ji}^{im} O_i^{in} \\ O_j^{im}(k) = f(Net_j^{im}(k)) \end{cases}, i = 1, 2, 3, 4, 5 \quad (6)$$

The activation function of hidden layer neurons is hyperbolic tangent function:

$$f(x) = \frac{e^x - e^{-x}}{e^x + e^{-x}} \quad (7)$$

The inputs and outputs of the output layer are:

$$\begin{cases} Net_k^{out}(k) = \sum_{j=1}^4 w_{kj}^{out} O_j^{im} \\ O_k^{out}(k) = g(Net_k^{out}(k)) \times A_k \end{cases}, k = 1, 2, 3 \quad (8)$$

The 3 output nodes of the output layer correspond to 3 ESO parameters  $\beta_1, \beta_2, \beta_3$  respectively, and  $A_k$  is the output layer coefficient, and the output layer neuron activation function uses the non-negative sigmoid function:

$$g(x) = \frac{e^x}{e^x + e^{-x}} \quad (9)$$

Defined the cost function of the system is:

$$\varepsilon(k) = \frac{1}{2} (rin(k) - yout(k))^2 \quad (10)$$

The application of back-propagation algorithm of BP neural network is very proven. So we don't give too much detail.

### C. Output Layer Coefficient Selection Method

This paper proposes a method. Multiply the output layer activation function outputs and the corresponding output layer

coefficients trained by the BP neural network, and then use it as the ESO parameter:

Firstly, the single parametric design method based on bandwidth concept which is proposed by Dr.Gao Zhiqiang is used to select the base values of a set of initial coefficients  $[3\omega, 3\omega^2, \omega^3]$  [6].  $\omega$  is the parameter related to system bandwidth, and the greater the value of  $\omega$ , the higher the bandwidth of the ESO

Because output value of the sigmoid function is non-negative and is less than or equal to 1. Therefore, the initial coefficient should be multiplied by a factor k greater than 1. Then, after the BP neural network is trained in the system, a set of parameters  $[A_1, A_2, A_3]$  of convergence is obtained. With this group of parameters multiplied by K as the initial factor, the training is performed again, and the value of X can be adjusted in training until the convergence speed of training is faster.

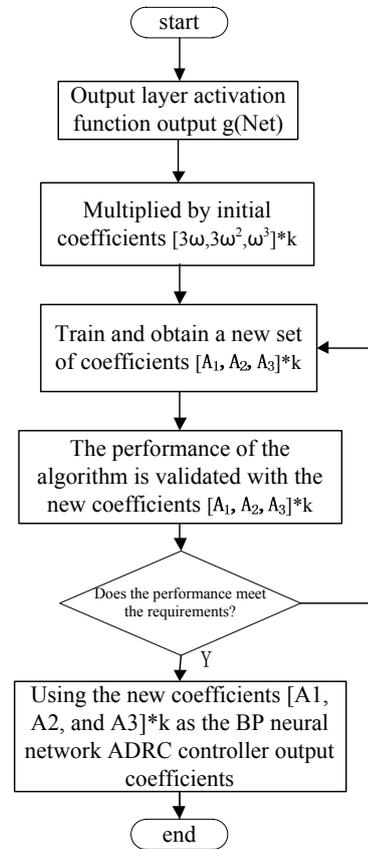


FIGURE III. INTERNAL STRUCTURE OF THE BP NEURAL NETWORK.

## IV. SIMULATION ANALYSIS

### A. Photoelectric Stabilized Platform Configuration

Photoelectric stabilized platform servo control system uses gyroscope, resolver encoder, image tracker etc. as the feedback element, the controller input signal processing and feedback signal and drives the motor to control the load platform. The controller processes the input signal, the feedback signal and drives the motor to control the load of the platform. The controller adjusts PWM wave signal to control the motor, so as

to adjust the direction of LOS. Gyro sense the LOS speed in inertial space, the encoder sense the angle of the mirror relative to the carrier, the image tracker output target miss distance. The schematic diagram of the working principle is shown in the following figure.

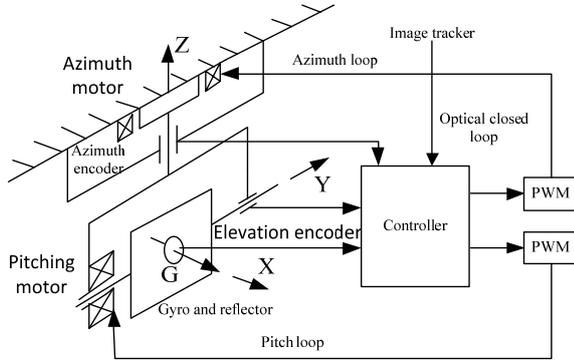


FIGURE IV. DIAGRAM OF A DOUBLE-AXIS STABILIZED PLATFORM OPERATING PRINCIPLE

The flow chart of the servo system of the single axis stabilization loop is shown in the following figure:

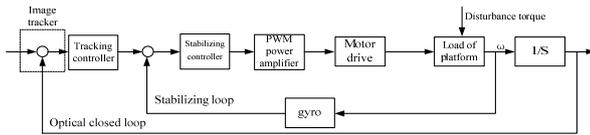


FIGURE V. DIAGRAM OF A SINGLE-AXIS STABILIZATION LOOP

Firstly, determine the controlled object model used for simulation. According to the parameters of the common airborne photoelectric stabilized platform, the model of the system is established:

$$\frac{1.25}{0.001034s^2 + 1.804s + 1.5625} \quad (11)$$

The disturbance model of the system is established according to the common disturbance parameters in the stability test.

### B. Disturbance Analysis and Modeling

When the carrier is moving in inertial space, the velocity disturbance in the inertial space will be generated. The disturbance is usually simulated by the sinusoidal disturbance generated by the angular vibrating table. In simulation, this sinusoidal disturbance is equivalent to the "total disturbances", which is added to the voltage input. The sinusoidal voltage disturbance with the amplitude of 1V and frequency of 1.6Hz is added to the system, as shown in the following figure.

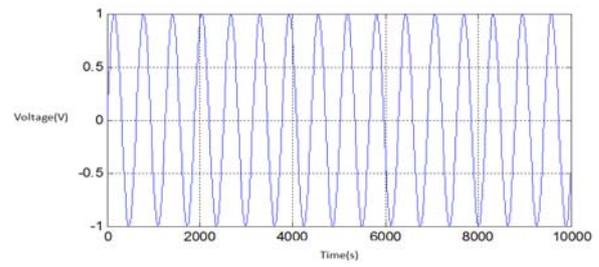


FIGURE VI. SINE VOLTAGE DISTURBANCE

Because of the friction torque between shafts, mass unbalance moment, winding torque and so on, there are disturbing torque single direction change in airborne photoelectric stabilized platform control system. Added voltage disturbance varies linearly with time  $w(k) = 0.0008 \times k + 1$  to simulate the "total disturbance" of these torques,  $k$  changes with time, every 0.001 seconds increased 1. In the simulation time of 10 seconds, the disturbance voltage increased from 1V to 9V, as shown in the following show.

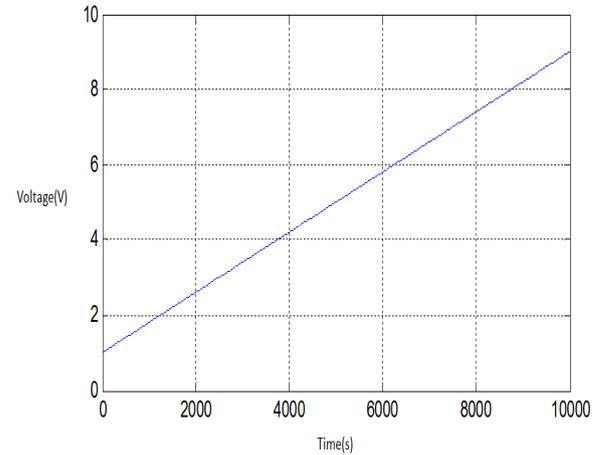


FIGURE VII. LINEAR VOLTAGE DISTURBANCE

### C. Simulation Methods and Results

In order to study the performance characteristics of ADRC based on BP neural network, this paper uses MATLAB to simulate the ADRC based on BP neural network in airborne photoelectric stabilized platform.

According to the method shown in Figure 3, first select a set of coefficients  $[3\omega, 3\omega^2, \omega^3] \times 2$  as the base value, including  $\omega=400$ , through repeated iteration to acquire a set of coefficients of  $[450 \ 120000 \ 18000000] \times 2$  as BP neural network output coefficients, the output values of the sigmoid function  $O_{k1}^{out}$ ,  $O_{k2}^{out}$ ,  $O_{k3}^{out}$ , multiplied by 900, 240000 and 18000000 respectively, and get  $[\beta_1, \beta_2, \beta_3]$  as ESO three parameters.

There are 2 parameters  $k_1, k_2$  of NLSEF control law need to be adjusted. We also used a method similar to ESO parameters tuning method to adjust the parameters of them. Inputs of the BP neural network are the system input, the

system output, the feedback error and the bias value 1, and the outputs are  $k_1$  and  $k_2$ .

In order to verify the disturbance isolation effect of ADRC based on BP neural network, the nonlinear ADRC is adopted as control.

The output amplitude contrast is shown in the figure below. Fig.(a) is the output of the ADRC. Fig.(b) is the output of the ADRC based on BP neural network. The blue line is the system input value, and the red line is the system output value.

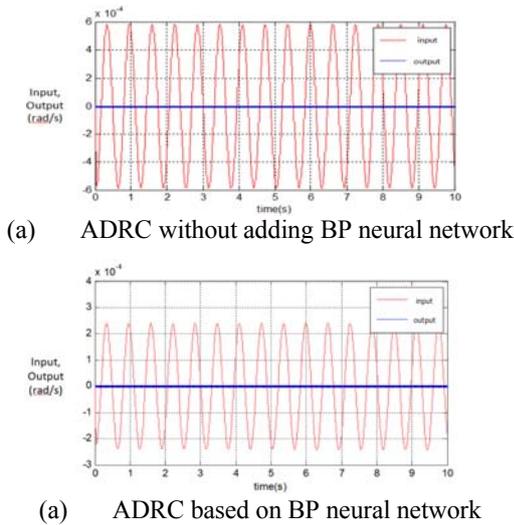


FIGURE VIII. STABILIZAIING ACCURACY CURVES WITH PLATFORM SINE DISTURBANCE(1)

Amplitude of Fig.(a) is about  $6 \times 10^{-4}$ rad/s, amplitude of Fig.(B) is about  $2.4 \times 10^{-4}$ rad/s, It can be seen that ADRC based on the BP neural network improved 1.5 times of sinusoidal disturbance isolation compared with traditional ADRC.

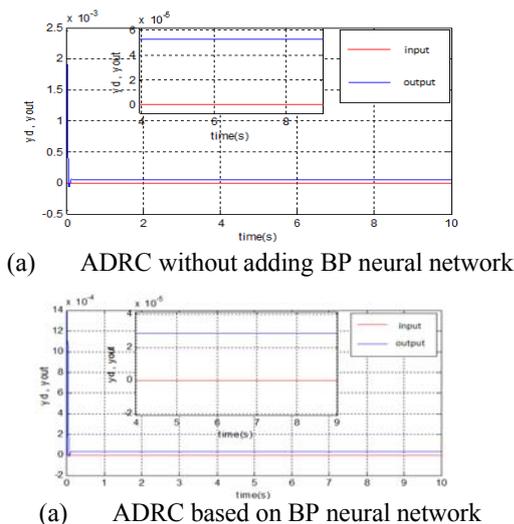


FIGURE IX. STABILIZING ACCURACY CURVES WITH PLATFORM SINE DISTURBANCE(2)

Peak value of Fig.(a) is about  $2.1 \times 10^{-3}$ rad/s, steady-state value of Fig.(a) is about  $5 \times 10^{-5}$ rad/s. Peak value of Fig.(b) is about  $1.4 \times 10^{-3}$ rad/s, steady-state value of Fig.(b) is about  $3 \times 10^{-5}$ rad/s, It can be seen that ADRC based on the BP neural network improved 67% of torque disturbance isolation compared with traditional ADRC.

## V. CONCLUSION

In this paper, the ADRC based on BP neural network in airborne photoelectric stabilized platform is studied, the stabilizing loop and disturbance are analyzed and modeled, the parameters tuning method of ADRC based on BP neural network parameters are analysed, the method to obtain the ESO parameter initial value is proposed. This method can effectively improve the convergence of BP neural network of ADRC. The ADRC based on BP neural network has been compared with the traditional ADRC in the airborne photoelectric stabilized platform through Simulation and Analysis.

The simulation results show that the parameters tuning method of the ADRC based on BP neural network can improve Disturbance isolation of airborne photoelectric stabilized platform significantly. This method has greatly improved the control effect has important significance to improve the LOS stabilization accuracy of airborne photoelectric stabilized platform.

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